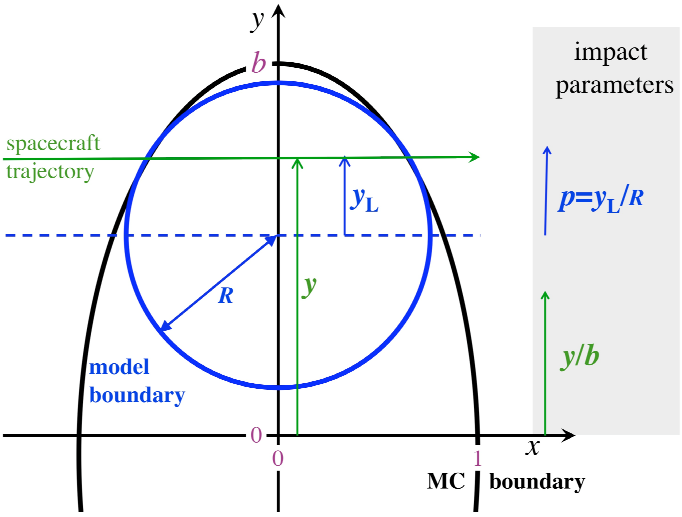
y
b
spacecraft
trajectory
$y_L$
R
y
model
boundary
0
0
1
x
MC
boundary
impact
parameters
$p=y_L/R$
y/b